# Transient Stabilization Improvement of Induction Generator Based Power System using Robust Integral Linear Quadratic Gaussian Approach

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Abstract- Interconnected power system consisting of a number of power generation units are able to fulfil the demand of electricity throughout the world. A safe and reliable operation of interconnected power system is possible when all the generators remain at synchronism state. Stability problem arises when the damping torque of these generators is reduced due to the effect of different uncertainties, faults and dynamic loads. The result of insufficient damping torque may deviate the speed of the generators and produce unsafe operation of power system due to the loss of stability and robustness of terminal voltage and rotor angle. In this paper, a novel integral linear-quadratic Gaussian (ILQG) controller is designed to regulate the oscillation of the power system for increasing stability and robustness. The better performance of the proposed controller is ensured by comparing it with the linear-quadratic regulator (LQR) and the linear-quadratic Gaussian (LQG) controllers. The comparison results ensure the optimum performance of the proposed controller against uncertainties and fault as compared to the LQR and LQG controller.

**Keywords:** Single machine infinite bus, Integral linear-quadratic Gaussian control, Speed regulation, Rotor angle regulation, Terminal voltage regulation.

## 1. Introduction

The electric demand over the world is increasing rapidly. A reliable and efficient solution to meet the growing demand of this electricity is the innovation of interconnected power system consisting of a number of generators [1], [2]. It provides a better and reliable supply of power and becomes exoteric with the development of the world [3], [4]. It receives energy from different sources and ensures reliability to the consumers. Wind turbine, solar, hydropower etc. are integrated to the main grid to provide additional energy throughout the year. It controls the system frequency and managing the tie transfer between utility regions and increases the reliability against disturbances [5]-[7].

The reliability and safe operation of the power system can be hampered due to fluctuation of the power. Power fluctuation occurs due to the unknown loads and faults [8][9]. The inertia and synchronising force may be hampered and system outages may be increased while operating the power system with renewable energies [10]. Again, the large demand of the electrical power may produce stress on the grid and hamper its stability [11]-[14]. The dynamics of the stressed power system is complex [15]. It exhibits nonlinear nature depending on the system structure, loads and unknown fault [16]. Although, the initial oscillation of the performances is quite small with the change of system loads, a large change of loads or disturbance may make it a disaster for the power system.

Induction machines and the components of power system also effect its performance. The machines of the interconnected power system must be synchronized for an unbreakable power supply [17]-[19]. Low damping torque, transmission line fault and unknown loads are responsible to hamper the synchronism of the machines and produce unsafe operation [20].

The synchronism of the power system are related to two control stability such as, (i) steady state and (ii) transient stability [21]. At small change of loads, power system stability is the static stability while the other one is responsible to synchronous all the machines of the power

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system. The transient stability is a function of both operational conditions and disturbances which is controlled by automatic voltage regulator (AVR) [22].

A weak control of the power system against different disturbances and faults increases the oscillations and deviations of its performances. The effects of the disturbances on the power system may increase the control challenge of rotor angle, speed and transient stability of the terminal voltage of the generators [23]. At synchronous state, any variation of the generator speed is minimized by damping torque to regain the synchronous speed of the generators. Dynamic loads, weak control of AVR, dynamic variation of the system parameters can reduce the restoring torque of the machine and cause instability [12].

The regulation of the damping torque of the generators to regain the operating equilibrium against different uncertainties is the main aim. A bounded system is able to withstand the disturbances and back to the equilibrium point. Controlling of each generator speed is the tendency to take the damping torque within equilibrium point to enhance the control performance of terminal voltage and rotor angle [24]-[27].

A number of control techniques have been proposed for a better control of different deviations of power system by controlling damping torque of the generators. Proportionalplus-Integral-plus-Derivative (PID) is the largely used industrial controller has been designed to enhance the stability of the interconnected power system [18], [28]-[30]. The simple design and implementation of the PID controller efficiently reduces the deviations of speed and rotor angle of the generator. Low bandwidth and robustness against uncertainties are the limitations of the PID controller.

Power system stabilizer (PSS) technique minimizes the limitations of the PID controller. It provides better protection of the power system against different uncertainties by restoring it [31], [32]. This control algorithm is deigned based on the control principle of fuzzy logic and sliding mode controller. Alone the fuzzy logic controller minimizes the effect of uncertainty and hazy of the power system. The low robustness against un-modelled loads is the limitation of this control approach. Sliding mode controller samples the sensor output whose objective function is minimized by fuzzy logic controller that are responsible for robust performance against different uncertainties.

Static synchronous compensator (STATCOM) has been proposed to control the stability of the power system. This control algorithm damps out the low-frequency oscillation produced by uncertainties and loads faults. This controller is designed based on the seeker optimization algorithm (OA) [33]. This algorithm is used to optimize the compensator parameter to minimize the objective function. STATCOM is also designed based on the cultural algorithm (CA) to tune the stabilizer parameter [34]. This controller faces optimization problem. For proper voltage regulation, distributed static synchronous compensator (DSTATCOM) has been proposed [35] by linearizing the the system order based on partial feedback linearization approach.



#### Fig. 1. Illusion of SMIB Structure.

To overcome the limitation of STATCOM controller, unified power flow controller (UPFC) has been proposed [36] that is a FACT device to improve the damping torque, active and reactive power. This control approach is designed based on electromagnetic transient algorithm. UPFC is difficult to use for power system having a large number of generation units.

The design of model predictive controller (MPC) has been proposed to enhance the stability of power system [8], [37], [38]. This control approach is designed with UPFC for better controlling of active and reactive power and damping torque. It has the ability to predict the control output and provides optimal function for desired performance. This control algorithm exhibits high gain and phase margin. The bandwidth of MPC against different loads and uncertainties is quite low.

The control structure of linear-quadratic regulator (LQR) has been proposed to reduce the deviations of the performances and increase the stability of the power system [39]-[41]. The controller design is based on linearization approach of mean value theorem. It provides a fast and accurate response which is optimal with respect to operating cost. The cost function of this optimal controller is minimized by the control algorithm of Algebraic Riccati Equation (ARE). Its performance may be harmed by dynamic loads.

A novel integral linear-quadratic Gaussian (ILQG) controller design is presented in this work for deviation control of the performances of single machine infinite bus (SMIB). The motivation for designing the ILQG controller over the LOR and the LOG controller is its ability to track the reference signal and robustness against the plant uncertainty. The design of the controller is done by augmenting the plant dynamics with an integrator which reduces the oscillation and provides the fast transient response as compared to the LQR and the LQG controller. The proposed ILQG controller minimizes  $H_2$  norm of the objective function which results in high gain of the controller. The aim of this paper is to guarantee the stable operation of the SMIB for different uncertainties. The controller performances are tested using Matlab. The results ensure that the ILOG controller shows the minimum deviation for rotor angle, speed and terminal voltage which confirms the better performance as compared to the LQR and the LQG controller.

The remaining paper is arranged as follows: System modelling and controller design is presented in Section 2 and 3. The performance investigation of the controllers is carried out in Section 4. Section 5 contains the conclusion of this paper.

#### 2. Power System Modelling

An interconnected power system consists of a number of generators that are used to produce electricity from renewable and non-renewable energy source. A constant dynamic behavior of all generators is essential for safe and better performance. This large number of generation units and components increase the complexity of interconnected power system to investigate its performances. Single machine infinite bus (SMIB) is the possible solution to analysis the control performance of one generation unit of power system that exhibits similar performance of all generation units connected to the power system. The basic structure of SMIB is presented in Fig. 1. The capacity of the infinite bus is large enough i.e infinite whose voltage and frequency remain constant.

A set of mechanical and electrical model are used to represents the dynamics of SMIB [16], [17], [42]. The overall nonlinear mechanical model to represent the generator of SMIB can be written as,

$$\dot{\widetilde{\delta}} = \widetilde{\omega}_{0}(\widetilde{\omega} - l)$$

$$\dot{\widetilde{\omega}} = \frac{l}{2H}(-D\widetilde{\omega} - P_{M} + P_{E})$$
(1)

Where  $\underset{\delta}{\sim}$ , *H* and *D* represent the rotor angle, inertia and damping constant of the generator.  $\widetilde{\omega}_{o}$  and  $\widetilde{\omega}$  are the synchronous and rotor speed.  $P_M$  and  $P_E$  are the supplied mechanical power to and delivered electrical power from the generator.

The nonlinear electrical dynamics model representation for generator are,

$$\dot{\widetilde{E}}_{q} = \frac{1}{T_{d}} (-E_{q} + \widetilde{E}_{fd})$$

$$\dot{\widetilde{E}}_{fd} = \frac{1}{T_{a}} [K_{a} (V_{ref} - V_{t}) - \widetilde{E}_{fd}]$$
(2)

Where  $\widetilde{E}_{q}$ ,  $E_{q}$  and  $\widetilde{E}_{fd}$  are the transient voltage, generator voltage and exciter voltage of the generator.  $K_a$ and  $T_a$  are the exciter gain and time constant.  $V_{ref}$  and  $V_t$ are the obtained terminal voltage from and applied reference voltage to the generator. The algebraic equations for electrical model of the generator can be represented as

$$E_q = \frac{\hat{x}_d}{\hat{x}_d} \tilde{E}_q - (x_d - \tilde{x}_d) \frac{V_s}{\tilde{x}_d} \cos \delta$$

$$I_d = \frac{\tilde{E}_q}{\hat{x}_d} - \frac{V_s}{\tilde{x}_d} \cos \delta$$

$$I_q = \frac{V_s}{\hat{x}_d} \sin \delta$$

$$P_E = \frac{\tilde{E}_q V_s}{\hat{x}_d} \sin \delta$$

$$Q_E = \frac{\tilde{E}_q V_s}{\hat{x}_d} \cos \delta - \frac{V_s^2}{\hat{x}_d}$$

$$V_q = \tilde{E}_q - \tilde{x}_d I_d$$

$$V_d = \tilde{x}_d I_q$$

$$\overline{V}_t = \sqrt{\overline{V_d}^2 + \overline{V_q}^2}$$

$$\hat{x}_d = \tilde{x}_d + \tilde{x}_T + 0.5 \tilde{x}_L$$

$$\hat{x}_d = x_d + \tilde{x}_T + 0.5 \tilde{x}_L$$

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Where  $x_d$  and  $\tilde{x}_d$  are the reactance and transient reactance of the generator.  $\widetilde{x}_T$  and  $\widetilde{x}_L$  are the transformer and transmission line reactance.  $I_q$  and  $I_d$  are the generator current alone quadrature- and direct-axis.  $V_{\rm s}$  and  $Q_{E}$  are infinite bus voltage and reactive power.

Now from Eq. (1) and (2), the overall representation of SMIB can be written as,

$$\dot{\widetilde{\delta}} = \widetilde{\omega}_{0}(\widetilde{\omega} - l)$$

$$\dot{\widetilde{\omega}} = \frac{l}{2H}(-D\widetilde{\omega} - P_{M} + \frac{\widetilde{E}_{q}V_{s}}{\widehat{x}_{d}}\sin\delta)$$

$$\dot{\widetilde{E}}_{q} = \frac{l}{T_{d}}(-\frac{\widehat{x}_{d}}{\widehat{x}_{d}}\widetilde{E}_{q} + (x_{d} - \widetilde{x}_{d})\frac{V_{s}}{\widetilde{x}_{d}}\cos\delta + \widetilde{E}_{fd})$$

$$\dot{\widetilde{E}}_{fd} = \frac{l}{T_{a}}[K_{a}(V_{ref} - \sqrt{(\widetilde{E}_{q} - \widetilde{x}_{d}I_{d})^{2} + (\widetilde{x}_{d}I_{q})^{2}}) - \widetilde{E}_{fd}]$$
(3)

The nonlinear model of SMIB is difficult to analysis. it is essential to convert the nonlinear system into linear to make ease the analysis of the performance of SMIB. A nonlinear time-variant system can be written as,

$$\dot{\widetilde{x}} = h(\widetilde{x}, u, t)$$
  
 $\widetilde{y} = g(\widetilde{x}, u, t)$ 

Where  $\tilde{x}$ , *u* and  $\tilde{y}$  are the system state, input and output vector. h and g are the nonlinear input and output vector function of the system. When the system does not depend on the time is called time-invariant system that can be represented as

$$\begin{aligned} \dot{\widetilde{x}} &= h(\widetilde{x}, u) \\ \widetilde{y} &= g(\widetilde{x}, u) \end{aligned}$$
 (4)

For synchronism, let the initial state and input vector at equilibrium point are  $\tilde{x}_0$  and  $u_0$ , then,

$$\dot{\widetilde{x}}_0 = h(\widetilde{x}_0, u_0) = 0$$

Any deviation  $\Delta$  from the equilibrium point makes new state vector  $\tilde{x} = \dot{\tilde{x}}_0 + \Delta \tilde{x}$  and input vector  $u = \dot{u}_0 + \Delta u$ , that are responsible for new state of the system such as,

$$\dot{\widetilde{x}} = h[(\dot{\widetilde{x}}_0 + \Delta \widetilde{x}), (\dot{u}_0 + \Delta u)]$$

With this small deviation, Taylor's series can be used to represent a complete nonlinear system such as,

$$\begin{aligned} \dot{\tilde{x}}_{i} &= h_{i}(\tilde{x}_{0}, u_{0}) + \frac{\partial h_{i}(\tilde{x}_{0}, u_{0})}{\partial \tilde{x}_{1}} \Delta \tilde{x}_{1} + \frac{\partial h_{i}(\tilde{x}_{0}, u_{0})}{\partial \tilde{x}_{2}} \\ \Delta \tilde{x}_{2} + \dots + \frac{\partial h_{i}(\tilde{x}_{0}, u_{0})}{\partial \tilde{x}_{n}} \Delta \tilde{x}_{n} + \frac{\partial h_{i}(\tilde{x}_{0}, u_{0})}{\partial u_{1}} \Delta u_{1} + \\ \frac{\partial h_{i}(\tilde{x}_{0}, u_{0})}{\partial u_{2}} \Delta u_{2} + \dots + \frac{\partial h_{i}(\tilde{x}_{0}, u_{0})}{\partial u_{n}} \Delta u_{n} \end{aligned}$$
(5)

Where i = 1, 2, 3, ...n. Since  $f_i = (\tilde{x}_0, u_0) = 0$ , Eq. (5) be represented as

can be represented as,

$$\begin{split} \Delta \dot{\widetilde{x}}_{i} &= \frac{\partial h_{i}(\widetilde{x}_{0}, u_{0})}{\partial \widetilde{x}_{1}} \Delta \widetilde{x}_{1} + \frac{\partial h_{i}(\widetilde{x}_{0}, u_{0})}{\partial \widetilde{x}_{2}} \Delta \widetilde{x}_{2} + \dots \\ &+ \frac{\partial h_{i}(\widetilde{x}_{0}, u_{0})}{\partial \widetilde{x}_{n}} \Delta \widetilde{x}_{n} + \frac{\partial h_{i}(\widetilde{x}_{0}, u_{0})}{\partial u_{1}} \Delta u_{1} + \frac{\partial h_{i}(\widetilde{x}_{0}, u_{0})}{\partial u_{2}} \\ \Delta u_{2} + \dots + \frac{\partial h_{i}(\widetilde{x}_{0}, u_{0})}{\partial u_{n}} \Delta u_{n} \end{split}$$

and for the output,

$$\dot{y}_{j} = g_{j}(\tilde{x}_{0}, u_{0}) + \frac{\partial g_{j}(\tilde{x}_{0}, u_{0})}{\partial \tilde{x}_{1}} \Delta \tilde{x}_{1} + \frac{\partial g_{j}(\tilde{x}_{0}, u_{0})}{\partial \tilde{x}_{2}}$$

$$\Delta \tilde{x}_{2} + \ldots + \frac{\partial g_{j}(\tilde{x}_{0}, u_{0})}{\partial \tilde{x}_{n}} \Delta \tilde{x}_{n} + \frac{\partial g_{j}(\tilde{x}_{0}, u_{0})}{\partial u_{1}} \Delta u_{1} + \frac{\partial g_{j}(\tilde{x}_{0}, u_{0})}{\partial u_{2}} \Delta u_{2} + \ldots + \frac{\partial g_{j}(\tilde{x}_{0}, u_{0})}{\partial u_{n}} \Delta u_{n}$$
(6)

Where  $j = 1, 2, 3, \dots n$ . Since  $g_j = (\tilde{x}_0, u_0) = 0$ , Eq. (6) can be represented as,



Fig. 2. Bode Diagram of Open-loop System.

$$\begin{split} \Delta \dot{y}_{j} &= \frac{\partial g_{j}(\widetilde{x}_{0}, u_{0})}{\partial \widetilde{x}_{1}} \Delta \widetilde{x}_{1} + \frac{\partial g_{j}(\widetilde{x}_{0}, u_{0})}{\partial \widetilde{x}_{2}} \Delta \widetilde{x}_{2} + \dots \\ &+ \frac{\partial g_{j}(\widetilde{x}_{0}, u_{0})}{\partial \widetilde{x}_{n}} \Delta \widetilde{x}_{n} + \frac{\partial g_{j}(\widetilde{x}_{0}, u_{0})}{\partial u_{1}} \Delta u_{1} + \frac{\partial g_{j}(\widetilde{x}_{0}, u_{0})}{\partial u_{2}} \\ \Delta u_{2} + \dots + \frac{\partial g_{j}(\widetilde{x}_{0}, u_{0})}{\partial u_{n}} \Delta u_{n} \end{split}$$

Therefore, the linearized from of Eq. (4) can be represented as,

$$\Delta \widetilde{x} = A \Delta \widetilde{x} + B \Delta u$$
$$\Delta y = C \Delta \widetilde{x} + D \Delta u$$

Where,

$$A = \begin{bmatrix} \frac{\partial h_1}{\partial \widetilde{x}_1} & \cdots & \frac{\partial h_n}{\partial \widetilde{x}_n} \\ \vdots & \vdots & \vdots & \vdots \\ \frac{\partial h_n}{\partial \widetilde{x}_1} & \cdots & \frac{\partial h_n}{\partial \widetilde{x}_n} \end{bmatrix} \quad B = \begin{bmatrix} \frac{\partial h_1}{\partial u_1} & \cdots & \frac{\partial h_n}{\partial u_n} \\ \vdots & \vdots & \vdots & \vdots \\ \frac{\partial h_n}{\partial u_1} & \cdots & \frac{\partial h_n}{\partial u_n} \end{bmatrix}$$
$$C = \begin{bmatrix} \frac{\partial g_1}{\partial \widetilde{x}_1} & \cdots & \frac{\partial g_n}{\partial \widetilde{x}_n} \\ \vdots & \vdots & \vdots & \vdots \\ \frac{\partial g_n}{\partial \widetilde{x}_1} & \cdots & \frac{\partial g_n}{\partial \widetilde{x}_n} \end{bmatrix} D = \begin{bmatrix} \frac{\partial g_1}{\partial u_1} & \cdots & \frac{\partial g_n}{\partial u_n} \\ \vdots & \vdots & \vdots \\ \frac{\partial g_n}{\partial u_1} & \cdots & \frac{\partial g_n}{\partial u_n} \end{bmatrix}$$

Where A, B, C and D are the state, input, output and transition matrix of a system. There are four state  $\tilde{x}$  for SMIB that can be written as,

$$\widetilde{x} = \begin{bmatrix} \Delta \widetilde{\delta} & \Delta \widetilde{\omega} & \Delta \widetilde{E}_{q} & \Delta \widetilde{E}_{fd} \end{bmatrix}^{T}$$

Based on this technique, Eq. (3) can be linearized as,



Fig. 3. Closed-loop Structure of SMIB with ILQG Controller.



Where  $K_1 - K_6$  can be written as,

$$K_{1} = \frac{\widetilde{E}_{q}V_{s}\widetilde{x}_{d}}{\cos\delta} - \frac{V_{s}^{2}(x_{q} - \widetilde{x}_{d})}{\widehat{x}_{d}\widehat{x}_{d}}\cos 2\delta$$

$$K_{2} = \frac{V_{s}}{\widehat{x}_{d}}\sin\delta$$

$$K_{3} = \frac{\widehat{x}_{d}}{\widehat{x}_{d}}$$

$$K_{4} = \frac{V_{s}(x_{q} - \widetilde{x}_{d})}{\widehat{x}_{d}}\sin\delta$$

$$K_{5} = \frac{V_{s}x_{q}V_{td}}{\widehat{x}_{q}\widetilde{V}_{t}}\cos\delta - \frac{V_{s}\widetilde{x}_{d}V_{tq}}{\widehat{x}_{d}V_{t}}\sin\delta$$

$$K_{6} = \frac{x_{T}V_{tq}}{\widehat{x}_{d}\widetilde{V}_{t}}$$

Here,  $V_{id}$  and  $V_{iq}$  are the *d* and *q* component of the terminal voltage of the induction machine. The performance of the open-loop frequency domain response is reported in Fig. 2. The peak value of frequency domain is almost 40 dB. The bode diagram shows that the open-loop system has positive gain. According to the control theory, a positive gain system is unstable. The aim of our proposed controller is to make the system stable by reducing the peak value against uncertainties.

#### 3. Design of Proposed Controller

Fig. 3 represents the control structure of ILQG controller [43], [44]. Consider the plant dynamics:



Fig. 4. Block Diagram of Inverse Multiplicative Output Uncertainty

Table 1. Design Parameter Value

LQR Controller	$R = 1; Q = 1 \times 10^{-5}$
LQG Controller	$R = 1 \times 10^{-1}; Q = 1 \times 10^{-7};$ $\omega_n = 1 \times 10^{-3}; \omega_d = 1 \times 10^{-4}$
ILQG Controller	$\begin{aligned} \gamma_1 &= 1 \times 10^8; \gamma_2 = 1 \times 10^5; \gamma_3 = 1 \times 10^{-5} \\ ; q &= 1 \times 10^7; r = 1 \times 10^{-8}; \end{aligned}$

$$\dot{x}(t) = A_1 x(t) + B_1 u(t) + D_1 \omega(t)$$

$$y(t) = C_1 x(t) + D_2 \omega(t)$$
(7)

where *x*, *u*, *y* and  $\omega$  are the state vector, system input, measured output and Gaussian white noise disturbance respectively;  $A_1$ ,  $B_1$ , and  $C_1$  state the system, input and output matrix respectively; while  $D_1$  and  $D_2$  are the system noise matrices. The cost function for the LQG controller is,

$$J = \lim_{T \to +\infty} E\left[\frac{1}{T}\int_{0}^{T} (x(t)^{T}Qx(t) + u(t)^{T}Ru(t))dt\right]$$
(8)

Here, the symmetric weighting matrices are  $Q \ge 0$  and R > 0, and the expected value is *E*. In Eq. (8), the term  ${}_{x}{}^{T}Qx$  represents the minimization of the states and the term  ${}_{u}{}^{T}Ru$  describes the minimization of the size of control inputs for the system. An integral action is included that minimizes the error. The updated plant is,

$$\dot{x}_{f}(t) = A_{2}x_{f}(t) + B_{2}u(t) + B_{2}\omega_{1}$$

$$z_{f}(t) = C_{2}x_{f}(t) + v$$
(9)

Here,  $v = [\omega_2 \ \omega_3]^T$  is the augmented state,  $x_f = [x \ \int y dt]^T$  and  $z_f = [z_1 \ z_2]^T$  are the output vectors. For updated plant,





**Fig. 5:** Deviation of (a) Open-loop Rotor Angle, (b) Closed-loop Rotor Angle (c) Open-loop Speed, (d) Closed-loop Speed (e) Open-loop Terminal Voltage, (f) Closed-loop Terminal Voltage (g) Open-loop Exciter Voltage, (h) Closed-loop Exciter Voltage. Open-loop Response is Represented by Blue (-) Line. Green (-) Line, Magenta (-) Line and Red (- -) Dashed Line Represent the Closed-Loop Response of the LQG, LQR and ILQG Controller Respectively.



**Fig. 6:** Deviation of (a) Open-loop Rotor Angle, (b) Closed-loop Rotor Angle (c) Open-loop Speed, (d) Closed-loop Speed (e) Open-loop Terminal Voltage, (f) Closed-loop Terminal Voltage (g) Open-loop Exciter Voltage, (h) Closed-loop Exciter Voltage. Open-loop Response is Represented by Blue (-) Line. Green (-) Line, Magenta (-) Line and Red (- -) Dashed Line Represent the Closed-Loop Response of the LQG, LQR and ILQG Controller Respectively against Uncertainty.

Performance	ILQG	LQG	LQR
	Controller	Controller	Controller
Rotor Angle	0.0197	1.19	6.19
Deviation			
(s)			
Speed	0.0195	0.02	6.56
Deviation			
(s)			
Terminal	0.132	2.13	6.31
Voltage			
Deviation			
(s)			
Exciter	10.01	6.89	2.12
Voltage			
Deviation			
(s)			
$A_2 = \begin{bmatrix} A_1 & 0 \\ C_1 & 0 \end{bmatrix}, B_2 = \begin{bmatrix} B_1 \\ 0 \end{bmatrix} \text{ and } C_2 = \begin{bmatrix} C_1 & 0 \\ C_1 & I \end{bmatrix}$			

Table 2. Closed-loop Step Response Comparison

where  $A_1$ ,  $B_1$  and  $C_1$  are defined in Eq. (7). Now  $\omega_1$ is the mechanical noise with variance  $\gamma_1^2$ ,  $\omega_2$  is the sensor noise with variance  $\gamma_2^2$  and  $\omega_3$  is the integral output sensor noise. Both are assumed as Gaussian white noise. The updated cost function based on Eq. (9) can be represents as,

$$J = \lim_{T \to +\infty} E\{\frac{1}{T} \int_{0}^{T} (x \ (t)^{T} Qx(t) + f(y)^{T} Q_{I} f(y) + (t)^{T} Qy(t) + f(y)^{T} Q_{I} f(y) + (t)^{T} Ru(t))dt\}$$

where f(y) is state weighting matrices equal to  $\int_0^\tau y(\tau) d\tau$ and  $Q_t \ge 0$  is an integral state weighting matrices. Now, the ILQG controller can be given as,

$$\dot{\widetilde{x}}_{f}(t) = A_{2} \,\widetilde{x}_{f}(t) + B_{2}u(t) + K \,(z_{f} - C_{2} \,\widetilde{x}_{f})$$

$$u(t) = -L \,\widetilde{x}_{f} \,.$$
(10)

Here, *K* is the Kalman gain such as,  $K = P_K C_2^T R_K^{-1}$ Where  $P_K$  can be calculated using Riccati equation:

 $\widetilde{P}_k = A_2 P_K + P_K A_2^T - P_K C_2^T R_K^{-1} C_2 P_K + Q_K$ 

Here,  $Q_{k} \ge 0$  and  $R_{k} > 0$  are process and measurement noise matrices given by,

$$Q_{K} = \gamma_{1}^{2} \quad ; \quad R_{K} = \begin{bmatrix} \gamma_{2}^{2} & 0 \\ 0 & \gamma_{3}^{2} \end{bmatrix}$$

Table 3. Closed-loo	p Step	Response	Comparison	against
Uncertainty				

Performance	ILQG	LQG	LQR
	Controller	Controller	Controller
Rotor Angle	0.0183	0.33	6.14
Deviation			
(s)			
Speed	0.00968	3.12	6.58
Deviation			
(s)			
Terminal	0.126	0.354	6.35
Voltage			
Deviation			
(s)			
Exciter	12.3	12.5	2.06
Voltage			
Deviation			
(s)			

Where  $\gamma_1$  is the standard deviation related to  $\omega_1$ ,  $\gamma_2$ and  $\gamma_3$  are the standard deviation related to  $\omega_2$  and  $\omega_3$ respectively. The gain *L* in Eq. (10) is given by,

 $L = R_C^{-1} B_2 P_C$ 

where  $P_{c} \ge 0$  can be calculated using Riccati equation,

 $\widetilde{P}_C = A_2 P_C + P_C A_2^T - P_C C_2^T R_C^{-1} C_2 P_C + Q_C$ Here,  $R_C$  is the weighting matrix of the controller and

$$Q_C = C_2^T \begin{bmatrix} 1 & 0 \\ 0 & q \end{bmatrix} C_2.$$

The controller design parameters  $\gamma_1$ ,  $\gamma_2$ ,  $\gamma_3$ , r and q are adjusted to achieve desired performance.

## 4. Performance Evaluation

The performance of SMIB is evaluated in this section under the proposed, LQR and LQG controller. MATLAB simulation is used as a tool to exam the performance of the SMIB. The parameters value of ILQG, LQG and LQR controller are listed in Table 1. Fig. 5(a), 5(c), 5(e) and 5(g) represent the open-loop performances of SMIB i.e. the deviation of rotor angle, speed, terminal voltage and exciter voltage of the generator. These results indicate that the performances are deviated from the desired result for insufficient damping torque of the induction machine. These deviations are reduced by using ILQG, LQG and LQR controllers are shown in Fig. 5(b), 5(d), 5(f) and 5(h).

The performance of SMIB is also investigated against uncertainty. Fig. 4 represents multiplicative output uncertainty where G(s) represents system transfer function and W(s) or  $\delta(s)$  is the model variation. The selection of the value of W(s) or  $\delta(s)$  is in such way that can produce 25%

Table 4.	Highlights	of the ILQG	, LQG and	l LQR	Controller
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Name of the	Advantages	Limitations	
Controllers			
ILQG Controller	(i) High band-	-	
	width, (ii) Large		
	gain- and phase-		
	margin, (iii)		
	Robust, (iv)		
	Track reference		
	signal		
LQG Controller	(i) High band-	(i) Can not track	
	width,	reference signal	
LQR Controller	(i) High band-	(i) Can not track	
	width,	reference signal,	
		(ii) Low	
		operating region	

variation of the reference signal. The open-loop response against uncertainty is represented in Fig. 6(a), 6(c), 6(e) and 6(g). The controlled performances are shown in Fig. 6(b), 6(d), 6(f) and 6(h) using ILQG, LQG and LQR controller. The controllers reduce these deviations to almost zero and make the system stable and reliable.

The comparative analysis between ILQG, LQG and LQR is investigated in this section which is used to find most optimum control algorithm for SMIB. Table 2 represents the comparison of settling time of ILQG, LQG and LQR controller. The settling time for rotor angle, speed and terminal voltage deviation are 0.0197s, 0.0195s and 0.132s respectively that are much lower as compare to LQG and LQR controller. Again, Table 3 represents the comparison of these controllers against uncertainty. Table 2 and Table 3 are the evidence that the settling time for ILQG controller is much lower as compared to the LQG and LQR controller. The results ensure that the ILQG controller is the most optimum controller that provides most robust performance.

## 5. Conclusion

The stability analysis of interconnected power system is quite challenging due to its complex characteristics and the effect of dynamic loads and uncertainties. The damping torque is an important factor that regains the proper rotor angle and speed of the generators by synchronizing them. The presence of any fault or uncertainty are responsible to reduce this damping torque that hampers the stability and robustness of the power system. The design of novel ILQG controller is presented in this paper to control the performances of SMIB such as the rotor angle, speed and voltage deviation by increasing the damping torque. To ensure the better performance of the proposed controller, a comparative analysis is investigated between the ILQG, LQR and LQG controller. The ILQG controller is efficiently able to reduce the oscillations of the power system and help the system to regain sufficient dampind torque. The ILQG controller has settling time as 0.0197s, 0.0195s and 0.132s for rotor angle, speed and terminal voltage deviation without uncertainties which are much lower as compared to the LQG and LQR controller. Again, the deviation of these performances against uncertainty for ILQG controller are also smaller as compared to the LQG and LOR controller. The main limitation of the ILOG controller is that it can not properly control the exciter voltage deviation of SMIB. Table 4 summarizes these controller based on their advantage and limitation. The results and Table 4 ensure better control performance of the ILQG controller as compared to the LQR and the LQG controller that provides robust control of the SMIB against different uncertainties and fault.

#### 6. Appendix

6.1 Power System Parameters

Generator: M = 2H = 7,  $\tilde{\omega}_0 = 314$ , D = 0,  $x_d = 1.752 pu$ ,  $\tilde{x}_d = 0.45 pu$ ,  $T_d = 7.27 s$ .

Exciter:  $K_A = 200, T_A = 0.02, K_I = 0.8645, K_2 = 0.8645, K_3 = 0.3135, K_4 = 1.4190 K_5 = 0.1465, K_6 = 0.4168, T_1 = 0.155, T_2 = 0.034$ 

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